

# TESTING OF CERAMIC MATRIX COMPOSITES

## Challenges, Tools, Pitfalls, and Opportunities

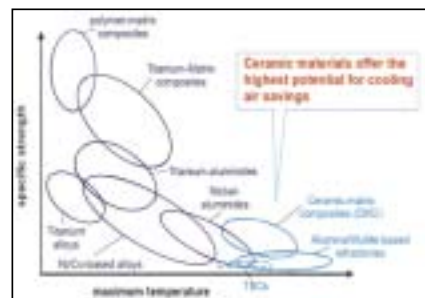
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**KEYWORDS:** ceramic matrix composite, mechanical testing, physical testing, high temperature, test standards

- Introduction
  - CMC Mechanical Properties
  - CMCs in Turbines and Nuclear Power
- CMC Material Challenges
  - Compared to Ceramic Monoliths
  - Compared to Polymer Matrix Composites
- Current ASTM C28 CMC Test Standards
  - Testing Objectives and Requirements
  - ASTM C28 CMC Tests
- Testing Issues in CMCs
- Need/Opportunities for New CMC Test Standards



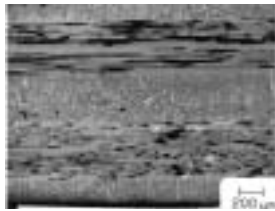
CMC Exhaust Seals in F414 Engine of F/A-18E/F (Super Hornet), The US Navy



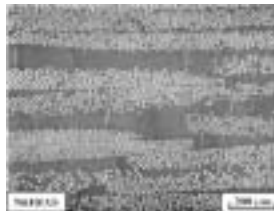
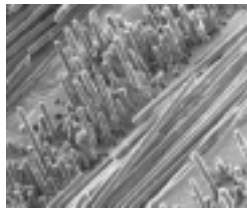
## **Fiber Reinforced Ceramic Matrix Composite –**

a ceramic matrix composite in which the reinforcing phase consists of a continuous fiber, continuous yarn, or a woven fabric.

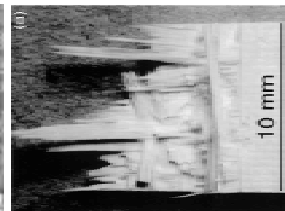
These components are combined on a macroscale to form a useful engineering material possessing certain properties or behavior not possessed by the individual constituents. (ASTM C1275)



Melt Infiltrated SiC-SiC Composite (GE)

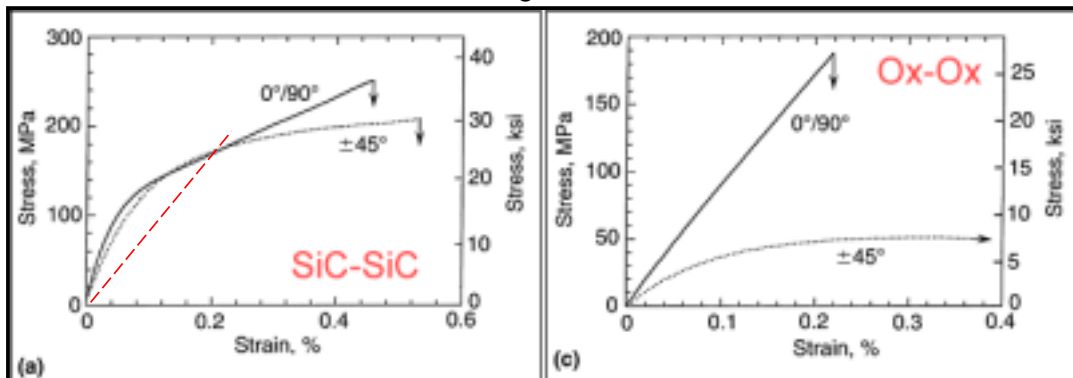


N610-Alumina-Silica Composite (Zawada, USAF)



**CMCs exhibit improved strain (~0.5%) damage tolerant failure from a cumulative damage process, unlike *monolithic* advanced ceramics that fracture catastrophically with low strain from a single dominant flaw.**

Tensile Stress-Strain Diagrams for Two CMCs



High Density, High Modulus MI SiC Matrix  
w/ Fiber Interface Coating on Nicalon (**WIC**)

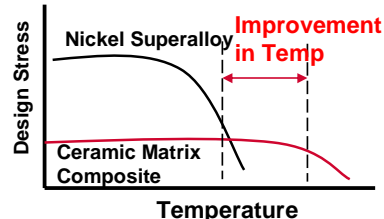
Porous, Low Modulus Al-Si-O Matrix  
No Fiber Coating on Nextel 720 (**WMC**)



## CMC Benefits --

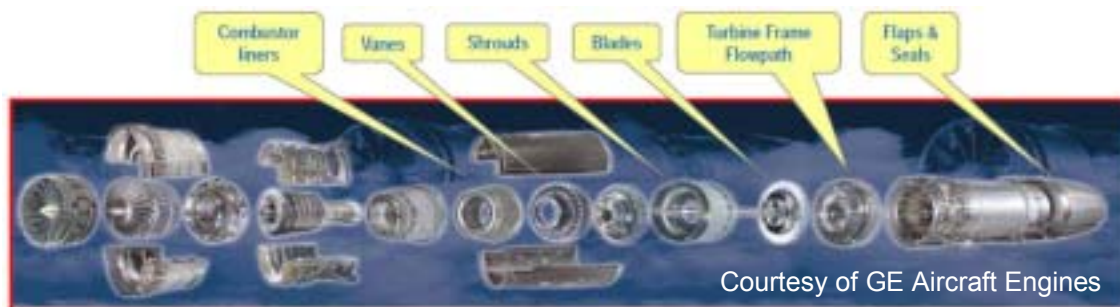
**High Temperatures** and **Lower Density** and **No Cooling** Compared to Metals → Fuel Efficiency

**Damage Tolerance** Compared to Ceramic Monoliths



## CMC Applications in Turbines --

Exhaust Components → Shrouds/Seals → Combustors → Stators/Vanes → Rotors/Blades



Courtesy of GE Aircraft Engines



## Oxide-Oxide CMC Nozzle and Center Body

90" Long, 40" Ø,

FAA CLEEN Program -- Boeing, ATK-COIC, AEC, Rolls Royce

Courtesy of FAA Cleen Program and Boeing



Courtesy of Aviation Week and GE

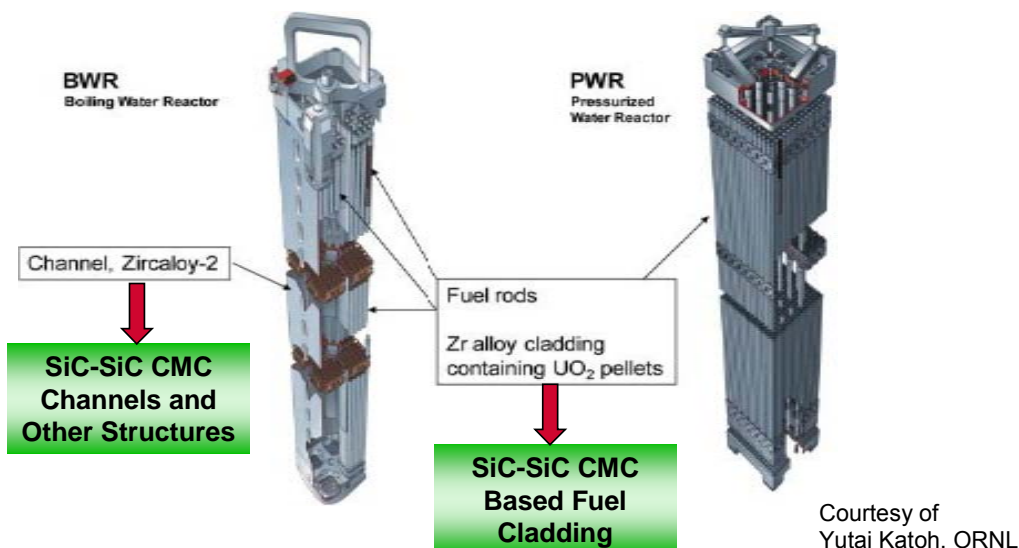
**Oxide-Oxide CMC**

**Mixer, Center Body, and Engine Core Cowl**  
on the GE Passport 20 engine  
16,500 # thrust

**MI SiC-SiC CMC**

**High Pressure Shrouds**  
in GE-CFM LEAP Engine  
30,000 # thrust

**CMC Challenges** -- Design Methodology, Cost, Reliability, HT Life, QA / NDE



**CMC Challenges** -- Cost, Reliability, HT Life, Design Methodology, QA / NDE



**Durability over Time and Temp in Variable Stress Conditions--**  
fatigue, impact, thermomechanical, erosion/wear/abrasion, thermal shock, etc

Life ↑ Time

**Environmental Durability over Time, Temperature,  
and Steady State Stress**

oxidation, creep/rupture, corrosion, radiation  
(Corrosion, Radiation) (Creep, Rupture, Corrosion, Radiation, Abrasion)

**Stress Concentration Effects**

**Mixed Stress States**

**Temperature Effects and Limits**

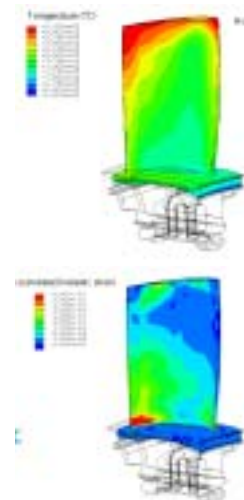
**Baseline/RT Mechanical and Physical Properties**

**An anisotropic composite consisting of ceramic fibers in  
a ceramic matrix with porosity and coatings (sometimes).**



## What does the designer need for his FEA and life models?

- Model the stress-strain relationships  
Elastic Stress-strain-- Modulus (Tension, Compression, Shear).  
Poisson's ratio  
Pseudo-ductile -  $f(\sigma, \epsilon)$
- Design limit stress and strains  
Prop. Limit stress and strain, Ultimate Stress and Strain
- Model the thermal load-temperature-thermal strain relationships  
Thermal Conductivity, Specific Heat, C-Thermal Exp.,
- Model stress concentration effects from features and flaws  
Feature/flaw geometry and size effects
- Predict life and reliability  
Temperature limits  
Temp-time-stress effects --Creep, fatigue, material degradation, flaw growth  
Oxidation, corrosion, environmental effects  
Damage and Failure Mechanisms



**Complication -- Consider fiber architecture  
and component anisotropy on directional  
properties**





## What does the production engineer need for product acceptance?

### Product and component tests to verify quality, condition, and properties

- Mechanical data
- Thermal data
- Physical data
  - density,
  - tolerances -- form, fit, features, finish
- NDE acceptance for flaws and defects



Complication -- Consider fiber architecture and component anisotropy on directional properties



## What do we want from our mechanical testing?

Test data that is accurate and repeatable and statistically valid for

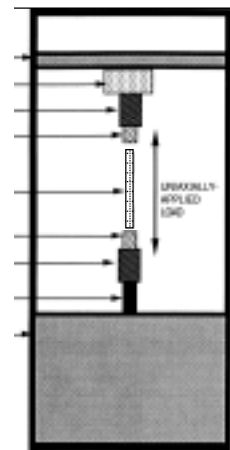
material development,  
material acceptance,  
QA and proof testing,  
reliability and life assessment

design data (FEA, allowables),  
product specifications,  
failure analysis.

Testing has to produce data that distinguishes between material variation and experimental variation.

And that means --

“Control and measure ALL the experimental variables and the material variables”



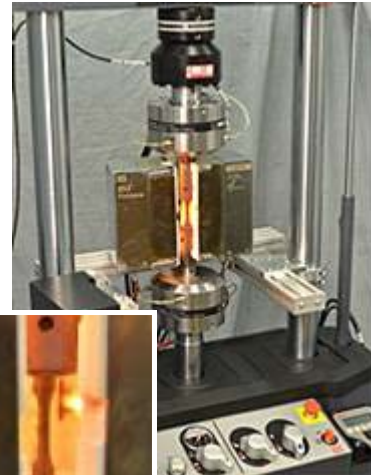
**Designers Need High Confidence, Statistically Proven Data**

## QUESTION -- Why can't we just use the SAME MECHANICAL TESTS that we use for ceramic monoliths and polymer matrix composite?

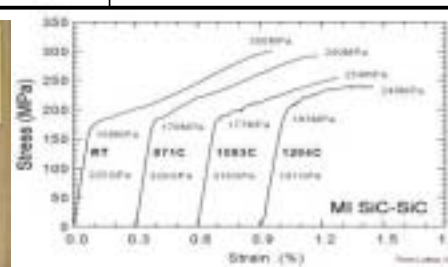
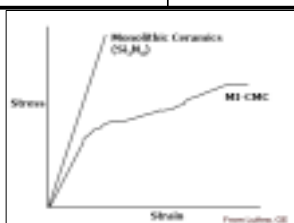
Because CMCs are different in their constituents and their properties. They differ in their stress-strain response and fail by different mechanisms than either ceramic monoliths or PMCs.

The temperature regimes for CMCs are much higher than for PMCs.

Tests have to be set up for the challenges of higher (~500-1500C) temperatures



Property	Ceramic Composite	Ceramic Monolith
Failure Mode	<b>Pseudo-ductile Failure with improved-strain, moderate strength,</b>	Elastic Low strain Brittle Failure, High strength
Critical Flaws	<b>Flaw tolerant</b>	Very flaw sensitive
Failure Mechanisms	Matrix failure → fiber damage → fiber failure	Brittle crack propagation in tension
Homogeneous??	Multiple components (fiber, matrix, interface coatings(?), porosity) combined at the macro level	One or two fine-grain components
Directional Properties?	<b>strong directional effects with fiber architecture and matrix vs fiber loading.</b>	Generally Isotropic
Temperature <b>HOT!!!</b> 500C- 1500C	Multiple Thermal/Oxidation Effects on Fibers, Matrix, Coatings	Thermal/Oxidation Effects on Material





Property	Ceramic Matrix Composite	Polymer Matrix Composite
Component Properties	Brittle, Cracked/ing Matrix w/ High Strength Fibers	Low Modulus, High Strain Matrix w/ High Strength Fibers
In-Plane 0° Failure Mechanisms	WIF -- Matrix failure → fiber damage → fiber failure WMF - fiber damage → fiber failure	Fiber Damage → Fiber Failure
Homogeneous ??	NO -- Multiple components (fiber, ceramic matrix, porosity/ cracks, interface coatings(?))	NO -- Two components – fiber and resin
Directional Properties?	Yes	Yes
Temp. Regime	500C up to 1500C	Up to ~300C



NASA SiC-SiC Combustor Liner



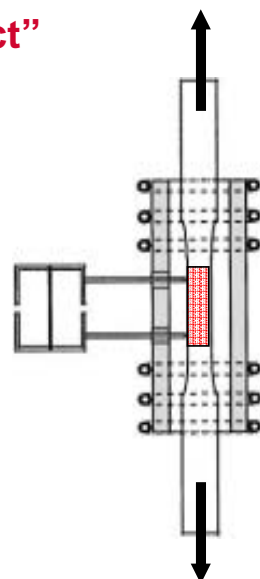
Japanese SiC-SiC Rocket Combustion Chamber

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## QUESTION – What do we need in our “perfect” mechanical tests?

- Accurate **force** application and measurement
- Accurate **strain** measurement in the gage section.
- **Consistent failure** in the gage section
  - **Uniform strain and/or stress** in the gage
  - **Maximum Stress** in the gage
  - **Uniform temperature** in the gage
  - **No uncontrolled stress concentrations or secondary stresses**
- **NDE and failure analysis** to assess material variation



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**ASTM Standard Test Method** -- a concise description of an orderly procedure for determining a property or constituent of a material, an assembly of materials, or a product.

- Includes all the essential details as to apparatus, test specimen, procedure, and calculations, and reporting needed to achieve satisfactory precision and bias.
- Represents a consensus as to the best currently available test procedure for the use intended
- Supported by experience and adequate data obtained from cooperative tests.

## General Outline

Scope

References

Terminology

Summary of Test Method

Significance and use

**Interferences**

Apparatus

Reagents and Materials

Hazards

**Test Specimens, Sampling**

Preparation of Apparatus

Calibration and Standardization

Conditioning

Procedure

Calculation or Interpretation of Results

**Report Requirements**

**Precision and Bias**

Keywords



## AT BEST!!!

- Confusion over terminology, test procedures, protocols, and data requirements.
- Conflicting and/or incomplete data comparison
- Multiple test requirements from multiple users and developers.



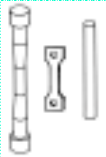
## AT WORST!!!

- **BAD TESTS** with uncontrolled variables giving invalid, unreliable, and incomplete test data.



## Monolithics

C 1161 Flexural strength  
C 1211 Flexural strength (High Temp)  
C 1368 Slow Crack Growth (Dynamic Fatigue)  
C 1465 Slow Crack Growth (High Temp)  
C 1576 Slow Crack Growth (Stress Rupture)  
C 1684 Flexural strength (Rods)



C 1424  
Compression  
strength



C 1421 Fracture  
Toughness



C 1322 Fractography  
C 1678 Fracture Mirror

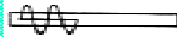


C 1326 Knoop hardness  
C 1327 Vickers hardness

C 1273 Tensile strength  
C 1366 Tensile strength (High T.)  
C 1291 Creep, Creep Rupture  
C 1361 Cyclic fatigue



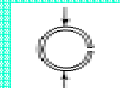
C 1499  
Biaxial strength



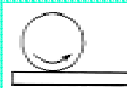
C 1198 Elastic Modulus  
- continuous  
C 1259 Elastic modulus  
- impulse



C 1470  
Thermal Guide



C 1323 C-ring strength



C 1495 Grinding



C 1525 Thermal shock

## Composites, Coatings, Porous Ceramics



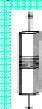
C 1275 CFCC Tensile strength  
C 1359 Tensile strength (Hi Temp)  
C 1337 Creep, Creep Rupture  
C 1360 Cyclic fatigue



C 1469 Joint  
strength



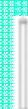
C 1341 CFCC Flexure strength  
C 1674 Honeycomb Flex  
strength



C 1358 CFCC  
Compression



C 1292 CFCC Shear strength  
C 1425 Shear strength (HiTemp)



C 1557  
Filament  
Tensile strength  
and  
Elastic modulus



C 1624 Coatings -  
Scratch Adhesion

## Powders



C 1070 Particle size, Laser Scatter  
C 1274 Particle Surface Area, BET  
C 1282 Particle size, Centrifugal Sed.



C 1494 C, N, O in silicon nitride

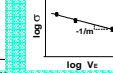
## Subcommittees

- .01 Mech. Prop. + Reliability
- .03 Physical Prop. + NDE
- .04 Applications
- .07 Ceramic Matrix Composites
- .91 Terminology

## NDE and Design



C 1239 Weibull  
Scaling



C 1683 Weibull  
Scaling



C 1175 NDE Guide



C 1212 Seeded voids  
C 1336 Seeded inclusions



C 1331 Ultrasonic velocity  
C 1332 Ultrasonic attenuation

## Terms, Workshops, Education



STP 1201 Life Prediction  
STP 1309 Composites  
STP 1392 Composites  
STP 1409 Fracture



C 1145 Terminology

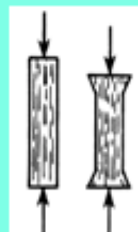


C 1275 CFCC Tensile strength

C 1359 Tensile strength (Hi Temp)

C 1337 Creep, Creep Rupture

C 1360 Cyclic fatigue



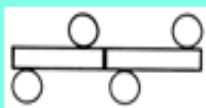
C 1358 CFCC  
Compression



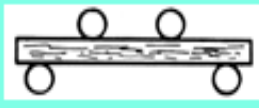
C 1468  
CFCC Tensile  
Trans  
thickness



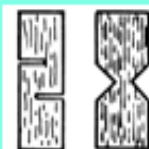
C 1557  
Filament  
Tensile strength  
and  
Elastic modulus



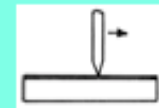
C 1469 Joint  
strength



C 1341 CFCC Flexure strength  
C 1674 Honeycomb Flex  
strength



C 1292 CFCC Shear strength  
C 1425 Shear strength (HiTemp)

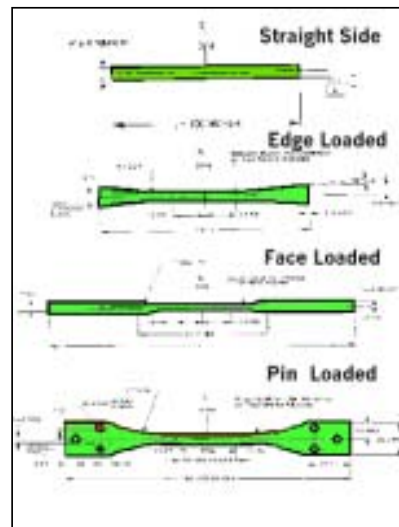


C 1624 Coatings -  
Scratch Adhesion



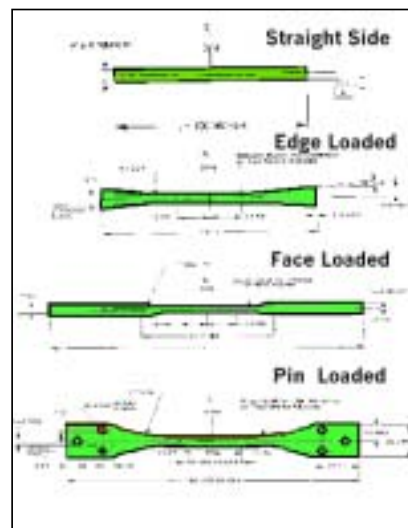
## C1275 / C1359 -- RT and HT **Tensile Strength** of CMC Flat Specimens C1773 - RT **Tensile Strength** of CMC Tubes (**New**)

- Flat Specimen Geometries
  - Straight Side with tabs
  - Edge Loaded - Reduced Gage Section
  - Face Loaded - Reduced Gage Section
  - Pin Loaded - Reduced Gage Section
- Key Experimental Variables
  - Test Rate** (Creep, Degradation, and Damage)
  - Size Effects and Gage Section Geometry**
  - Surface Condition**
- Interferences and Challenges
  - Out-of-Gage Failure**
    - Alignment and Bending Stresses**
    - Grip Stresses**
  - HT Strain Measurement**



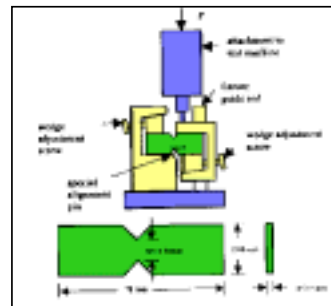
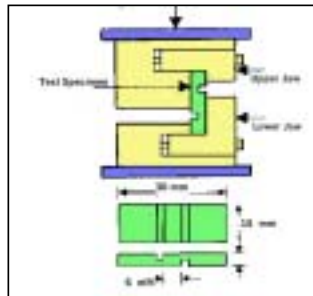
## ASTM C1337 – HT **Tensile Creep and Creep Rupture** of CMC Flat Specimens ASTM C1360 - RT **Tensile Fatigue** CMC Flat Specimens

- Flat Specimen Geometries
  - Same as Tensile Tests
- Key Experimental Variables
  - Test Rate** (Creep, Degradation, and Damage)
  - Size Effects and Gage Section Geometry**
  - Surface Condition**
  - Controlled and Uniform Temperature**
  - Atmosphere Monitoring and Control**
  - R ratio for fatigue**
- Interferences and Challenges
  - High Temperature Gripping**
  - Out-of-Gage Failure**
  - Alignment, Bending, and Grip Stresses**
  - HT Strain Measurement**



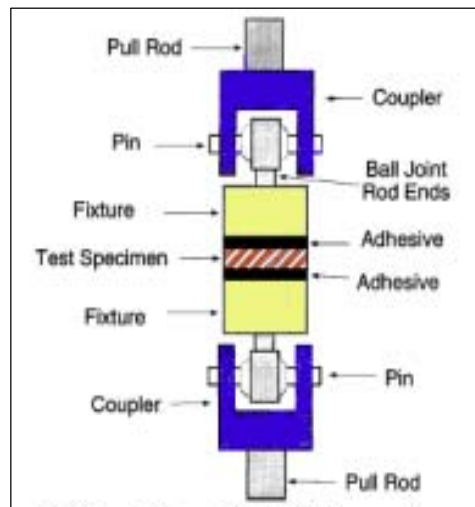
## ASTM C1292 and C1425 -- RT and HT *Shear* – CMC Flat Specimens

- Specimen Geometries
  - Double Notch Shear (Interlaminar)
  - Iosepescu ( In Plane)
- Key Variables
  - Notch Geometry and Specimen Size
- Interferences and Challenges
  - **Notch Stress Concentrations**
  - **Alignment and Mixed Mode Stresses**
  - **No Easy Strain Measurement**
  - Fixture Limits on HT testing



## ASTM C1468 -- RT *Transthickness Tensile* – CMC Flat Specimens

- Specimen Geometries
  - Round and Square
- Key Variables
  - Specimen Size
- Interferences and Challenges
  - **Epoxy Failure**
  - **Epoxy Limits the HT testing**
  - **Edge Effects**
  - **Bending Stresses**
  - **No Easy Strain Measurement**



## ASTM C1358 -- RT **Compression** -- CMC Flat Specimens

### Specimen Geometries

- Straight with Tabs
- Tapered Dog-Bone

### Key Variables

- Specimen Size and Flatness
- Alignment and Bending Stresses
- Specimen Anti-Buckling Fixture

### Interferences and Challenges

- Out of Gage Failures
- Buckling Failure

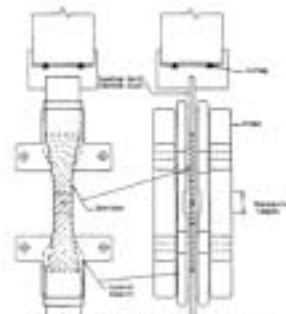
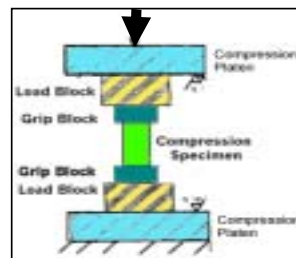


Fig. 1 Example of a Combined Face-Supported Fixture (1)

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## ASTM C1341 -- RT and HT **Flexure** -- CMC Bars

### Specimen and Loading Geometries

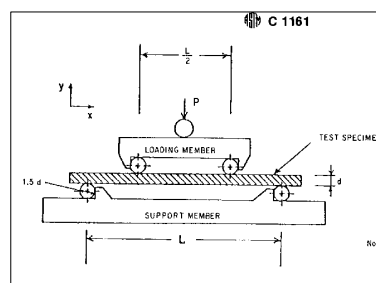
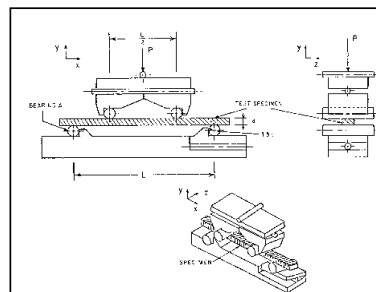
- 3-Point and 4-Point Loading
- Semi and Fully Articulating Fixtures

### Key Variables

- Flatness and Twist in Specimens
- Specimen Size and Geometry
- Span-to-Depth Ratio

### Interferences and Challenges

- Bending Stress vs True Tensile Face Stress (Elastic Beam Equations)
- Size Effects
- Invalid Failures (Shear, O-Gage, Contact)
- Mixed Mode Stresses and Surface Damage
- Temperature Variation in the Gage



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## ASTM C 1557 - RT *Tensile Strength and Young's Modulus of Fibers*

### Specimen Geometries

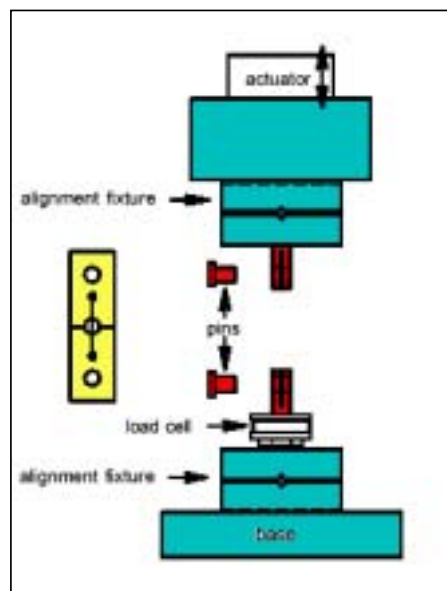
- Single Filament in Paper Tabs

### Key Variables

- **Specimen Length**
- **Strain by Flags or Calculation**

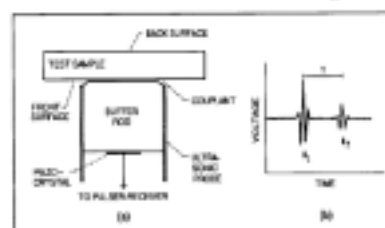
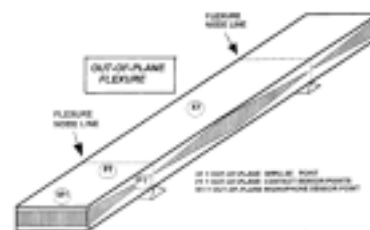
### Interferences and Challenges

- **Alignment and Bending Stresses**
- **Test Rate and SC Growth from Moisture**
- **Surface Damage**
- **Measurement of Filament Cross-Section Area**



## MODULUS AND ULTRASONIC TESTING

- **C1198 Dynamic Young's Modulus, Shear Modulus, and Poisson's Ratio** for Advanced Ceramics by **Sonic Resonance**
- **C1259 Dynamic Young's Modulus, Shear Modulus, and Poisson's Ratio** for Advanced Ceramics by **Impulse Excitation** of Vibrations
- **C1331 Ultrasonic Velocity** in Advanced Ceramics with the Broadband Pulse Echo Cross Correlation Method
- **C1332 Ultrasonic Attenuation** Coefficients of Advanced Ceramics by the Pulse Echo Contact Technique



## CMC Issue – Anisotropic Properties and Heterogenous Structure

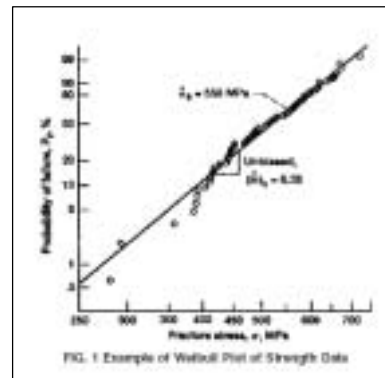
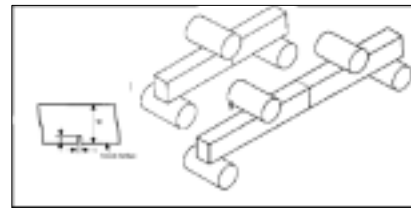


## MECHANICAL TESTS AND DATA ANALYSIS (Monolithic)

- **C 1421 Fracture Toughness** of Ceramics (SENB)

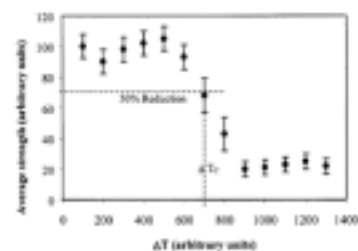
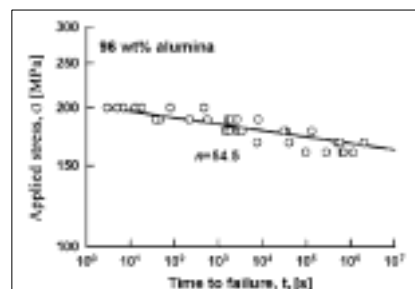
**CMC Issue –  $K_{IC}$  versus Crack Growth Resistance (Strain Energy Release Rate)**

- **C1239** Reporting Uniaxial Strength Data and Estimating **Weibull Distribution Parameters** for Advanced Ceramics
- **C1683 Size Scaling of Tensile Strengths Using Weibull Statistics** for Advanced Ceramics



## DURABILITY TESTS (Monolithic)

- **C1368 Slow Crack Growth** Parameters of Advanced Ceramics by **Constant Stress Rate Flexural** Testing at **Ambient Temperature**
- **C1465 Slow Crack Growth** Parameters of Advanced Ceramics by **Constant Stress Rate Flexural** Testing at **Elevated Temperature**
- **C1576 Slow Crack Growth** Parameters of Advanced Ceramics by **Constant Stress Flexural** Testing (Stress Rupture) at Ambient Temperature
- **C1525 Thermal Shock Resistance** for Advanced Ceramics by Water Quenching

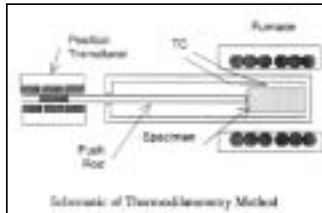




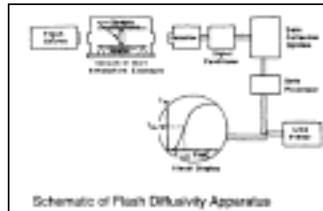
## THERMAL AND PHYSICAL TESTS (Monolithic)

- **C1470** Guide for Testing the **Thermal Properties** of Advanced Ceramics

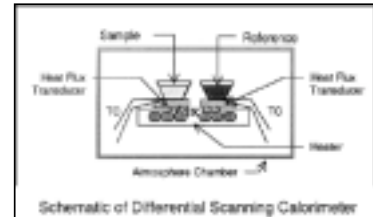
Coef. Thermal Expansion



Thermal Diffusivity/Conductivity



Specific Heat

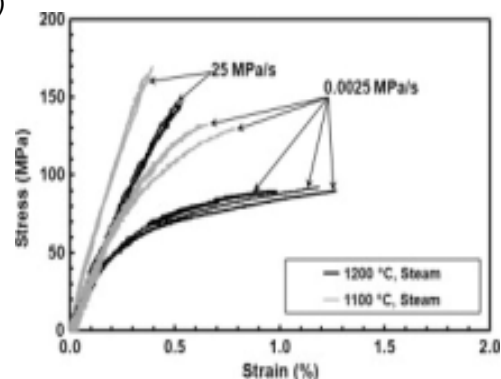


- **C1274** -- **Specific Surface Area by Physical Adsorption**
- **C1039** – **Density and Porosity by Archimedes Method**
- **D4284** - **Mercury Porosimetry**



## CMC Material and Challenges

- **Anisotropic Properties and Heterogeneous Structure**
- Material Variability (Structure and Defects)
  - In-specimen, between specimen, lot-to-lot
  - Internal Porosity, Flaws, and Defects
  - NDE of Test Specimens
- Specimen Size Effects  
(Architecture and Processing Effects)
- Hole, Notch, and Edge Stresses
- Surface Flaws from Spec Prep
- **Temperature, Strain Rate, and Atmosphere Effects (Creep and SC Growth)**
- Determining High Temp Failure/  
Degradation Mechanisms  
(Fiber, Matrix, Interface??)  
Intermediate Temp. Degradation



Steam, Temp and Strain Rate Effects on Tensile Strength of Oxide-Oxide Composite (N720, Alumina Matrix)

## High Temperature (>800C) Experimental Challenges

- Temperature Uniformity and Accurate Measurement
- Strain Measurement at HT (Accurate and Representative)
- High Temperature Gripping and Loading Fixtures

(Alignment/Ext.Stresses, Stability)

- Complex Environment (Combined Stress, Temperature, Exhaust, Atmosphere, Cycles)



→ Burner Rig Testing (HiV, HiP, HiT)

NASA Burner Rig  
6 atm, 200m/s, 2500F

- Thermal Shock Effects versus Thermal Degradation

## Precision and Repeatability

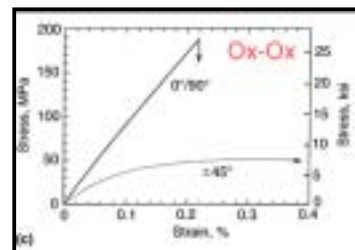
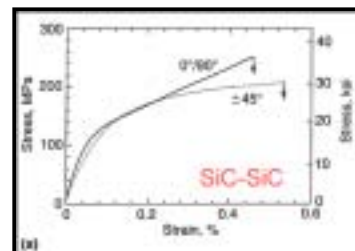
Key Question --What are the

**design limit factors,**  
per the designer's criteria?  
**Ultimate stress and strain or**  
**proportional limit stress and strain**

What is the 95% or 99%  
reliable stress level?

Do either of these strengths depend on  
**weakest link mechanisms** with  
probabilistic strength distributions,  
based on the **inherent variability in**  
**the composite:**

fibers, matrix, porosity, fiber interface coatings, surface seal coats  
fiber architecture, alignment, and anisotropy, inherent surface and volume flaws.

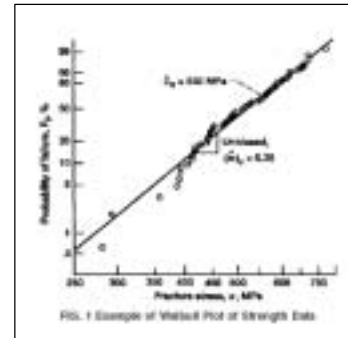


**When are Weibull statistics the best way to assess failure probability?**

## Where does the data variability come from??

### Material Variability occurs

spatially within a single test specimen,  
between test specimens,  
and between lots.



### Data variation also develops from

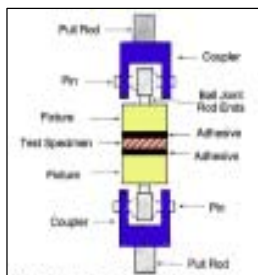
#### experimental variability –

test specimen dimensions and volume/size effects,  
extraneous bending stresses, notch effects  
slow crack growth, temperature and humidity effects

The tester and the data user need to know what the source of variability is in a given set of data.

## High Temp CMC Tests

- Interlaminar Tension
- Compression
- Thermal Gradient Stresses
- Fatigue (Tensile and Bending)



## Complex High Temp Test

- Thermocyclic Tests
- Oxidation and Corrosion (Stressed)
- Thermal Shock Damage
- Burner Rig Tests

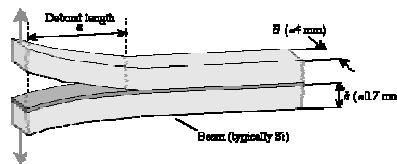






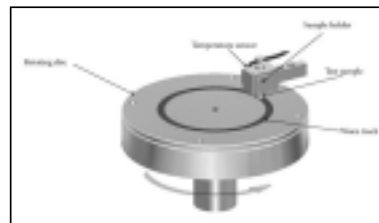
## Level 2 CMC Tests

- Crack Growth Resistance (Work of Fracture??)
- Open Hole and Notch Effects
- Mixed Stress State Tests



## Level 3 CMC Tests

- Impact Damage Tolerance
- Wear, Abrasion, Erosion
- Subelement Tests (Tubes, T's, L's)
- Strength of Joints
- Sandwich Structure Tests



## A Guide to NDE of CMCs



1. **Testing of CMCs is not simple, especially at high temperatures,** because of the range of material variables, properties, and fracture/damage mechanisms.
2. **ASTM C28 has many of the baseline mechanical property test standards.**
3. **There is a need for more high temperature and Level 2 and Level 3 test standards.**

*If you and your market need performance-oriented ceramic test standards to stay competitive and move your ceramic technology forward, use C28 standards and join us on the C28 committee!!*

[www.astm.org/COMMITTEE/C28.htm](http://www.astm.org/COMMITTEE/C28.htm)